

Europäisches Patentamt
European Patent Office
Office européen des brevets



(11) EP 0 838 394 A2

(12) EUROPEAN PATENT APPLICATION

(43) Date of publication:
29.04.1998 Bulletin 1998/18

(51) Int. Cl.⁶: B64C 9/02, B64C 9/16

(21) Application number: 97203214.8

(22) Date of filing: 15.10.1997

(84) Designated Contracting States:
AT BE CH DE DK ES FI FR GB GR IE IT LI LU MC
NL PT SE
Designated Extension States:
AL LT LV RO SI

(72) Inventor: Siers, Geoffrey L.
Kent, Washinton 98032 (US)

(74) Representative:
Land, Addick Adrianus Gosling
Arnold & Siedsma,
Advocaten en Octrooigemachtigden,
Sweelinckplein 1
2517 GK Den Haag (NL)

(30) Priority: 22.10.1996 US 735233

(71) Applicant: THE BOEING COMPANY
Seattle, Washington 98124-2207 (US)

(54) Slotted cruise trailing edge flap

(57) A slotted cruise trailing edge flap (16) having a single-slotted configuration during cruise, and a double-slotted configuration during takeoff and landing is disclosed. The slotted cruise trailing edge flap (16) includes a flap assembly (18) positioned along the trailing edge of an aircraft wing (30). The flap assembly (18) includes a vane element (22) and a main element (24) mounted in a fixed relation to one another that defines a slot (25) therebetween. The slotted cruise trailing edge flap also includes an extension assembly (20) that includes a support structure (23) from which the flap assembly (18) is translatable connected via a track (44) and carriage (46) arrangement. The flap assembly (18)

is rotatably connected to the support structure (23) using a cam follower (52) and cam slot (50). An actuation mechanism (28) including a rotary actuator (56) moves the flap assembly (18) relative to the support structure (23). In a stowed position, the vane element (22) preferably nests into the wing (30) such that the slot (25) remains available to direct airflow from the wing lower surface to the wing upper surface. In an extended position, the vane and main elements (22), (24) form a double-slotted arrangement created by the flap assembly translating rearward and rotating downward relative to the wing (30).

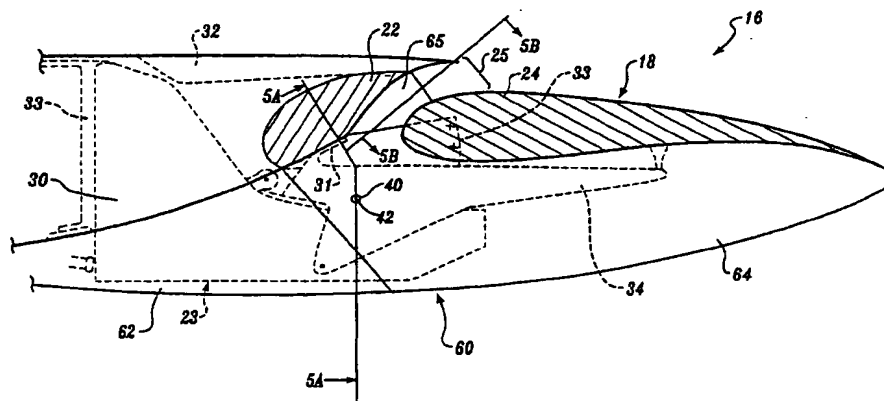


Fig. 1

EP 0 838 394 A2

EV 298497300 US

Description

The present invention relates to flap arrangements for aircraft wings, and more particularly, to double-slotted trailing edge flaps.

Various types of single- and multiple-slotted flaps are currently used during takeoff and landing to increase lift and improve aircraft maneuverability. During cruise, however, slots were previously thought to hold no value. Instead, designers have consistently tried to minimize the amount of airflow interruption experienced along the wing upper surface, particularly at the seam created between the flaps and the wing trailing edge or spoilers. See, for example, U.S. Patent No. 4,120,470.

It has recently become known that a spanwise slot of suitable contour positioned along the wing trailing edge can provide an aerodynamic benefit in the cruise regime by shaping the slot in such a way as to promote a controlled airflow from the lower to the upper surface of the wing. The addition of air from under the wing can actually re-energize the boundary layer on top of the wing. This results in a decrease in drag for a given airspeed. It is possible to trade this benefit for a decreased wing sweep with a possible resultant decrease in weight. A leading article regarding this phenomenon is Design and Optimization Method for Multi-Element Airfoils; Paper No. 93-0969, by Mark Drela, presented at the Aerospace Design Conference (Feb. 16-19, 1993) sponsored by the American Institute of Aeronautics and Astronautics.

In accordance with the present invention, a trailing edge flap having a single-slotted configuration during cruise and a double-slotted configuration during takeoff and landing is provided. The trailing edge flap includes a flap assembly that includes a vane airfoil element and a main airfoil element. The vane and main elements have a fixed relation to one another that defines a permanent slot therebetween. In a stowed position, the vane element nests into the wing such that the slot remains available to direct airflow from the wing lower surface to the wing upper surface. In an extended position, the vane and main elements translate rearward and rotate downward causing a second slot to form between the vane element and the wing trailing edge. Alternatively, the present invention may be formed to avoid the formation of the second slot by simply rotating the flap assembly downward without translating aft.

In accordance with further aspects of this invention, the flap assembly is supported at multiple lateral support locations positioned along the wing trailing edge. Each support location includes an extension assembly for moving the vane and main elements between stowed and extended positions. The flap assembly further includes a pallet having an upper surface and opposed pallet sidewalls. Pallets are provided at each support location. The vane and main airfoil elements are mounted to the pallet upper surface. A central pivot

axis extends between the pallet sidewalls in a lateral direction. The extension assembly includes a support structure from which the pallets are translatably and rotatably connected. In particular, a frame is provided having side panels oriented in a generally vertical-longitudinal plane and cantilevered from the wing rear spar. In the stowed position, the pallet sidewalls are positioned closely adjacent the outer surfaces of the frame side panels, thus forming a sleeve-type arrangement. An outer fairing is attached to the wing to surround portions of the extension assembly at each support location.

In accordance with other aspects of this invention, the support structure further includes a carriage translatably engaged with a track. The track is attached to the frame between the frame side panels. The carriage includes at least one wheel located below the track and at least one wheel located above the track. The carriage is rotatably connected to the pallet, preferably, by locating the central pivot axis coextensive with one of the carriage wheel axes. The support structure further includes a cam slot located near a lower region of each frame side panel outer surface. A cam follower formed in each pallet sidewall interior surface engages its respective frame side panel cam slot. The flap assembly may be made to follow a unique deflection path by altering the shape and position of the cam and track arrangements. In a preferred embodiment, the track and cam slot are shaped to cause the pallet to first translate aftward and then rotate downward in going from a stowed position to a fully extended position.

In accordance with still other aspects of this invention, an actuation mechanism is provided that includes a rotary actuator linked to the pallet to cause the flap assembly to rotate about the central pivot axis. In a preferred embodiment, the rotary actuator turns a radial arm that is pivotably connected to one end of an idler link. An opposite end of the idler link is pivotably connected to an extension tab located at a forward region of the pallet near the pallet upper surface.

The foregoing aspects and many of the attendant advantages of this invention will become more readily appreciated as the same becomes better understood by reference to the following detailed description, when taken in conjunction with the accompanying drawings, wherein:

FIGURE 1 is a simplified cross-sectional side view of a slotted cruise trailing edge flap formed in accordance with the present invention shown in a stowed position;

FIGURE 2 is a cross-sectional side view of the slotted cruise trailing edge flap of FIGURE 1 shown in a partially extended position;

FIGURE 3 is a cross-sectional side view of the slotted cruise trailing edge flap of FIGURE 1 shown in a fully extended position;

FIGURE 4 is a cross-sectional side view along line

4-4 of FIGURE 2 showing the idler link included in the slotted cruise trailing edge flap shown in FIGURES 1-3;

FIGURE 5A is a cross-sectional view along line 5A-5A of FIGURE 1 showing the carriage assembly included in the slotted cruise trailing edge flap shown in FIGURES 1-3; and

FIGURE 5B is a partial cross-sectional view along line 5B-5B of FIGURE 1 showing portions of the outer fairing included in the slotted cruise trailing edge flap shown in FIGURES 1-3.

As will be known to those skilled in the art of airplane wing design, an airplane wing 30 includes a number of rigid spars that extend the lateral (i.e., spanwise) distance of the wing. Ribs interconnect the spars to provide longitudinal structural support. Upper and lower skins are attached to the ribs and spars to close out the wing and to form a smooth, aerodynamic airflow surface. Of particular interest to the present invention is the wing rear spar 31 shown schematically in FIGURES 1-3 in combination with the rear fragment of a wing 30. The components of the present invention are generally located aft of, and are structurally supported by, the wing rear spar 31.

In general, a slotted cruise trailing edge flap 16 formed in accordance with the present invention has a single-slotted configuration during cruise, and a double-slotted configuration during takeoff and landing. This is accomplished by a flap assembly 18 that is movable between a stowed position and an extended position. In the stowed position a single slot is present, and in the extended position two slots are present. More specifically, flap assembly 18 includes two airfoil elements, a vane element 22 and a main element 24, that are arranged in fixed relation to one another. The space between the airfoil elements forms a permanent single slot 25. At various support locations along the wing trailing edge, the flap assembly 18 is movably connected to an extension assembly 20 that is secured to the wing rear spar 31. The extension assembly 20 includes a support structure 23 to which the flap assembly 18 is translatably and rotatably connected. The extension assembly 20 further includes an actuation mechanism 28 that moves the flap assembly 18 relative to the support structure 23. In a stowed position, the vane flap 22 nests into the wing 30 such that the permanent single slot 25 remains available to direct airflow from regions below the wing to regions above the wing. In an extended position, the vane and main elements 22, 24 form a double-slotted arrangement by rotating downward and translating rearward relative to the wing 30.

In more detail, the vane and main airfoil elements 22, 24 are attached to the wing 30 at a number of spanwise support locations positioned at short intervals along the wing trailing edge. One embodiment of a support location is shown in the cross-sectional side views of FIGURES 1-3. As best seen in FIGURES 2 and 3, the

vane and main elements 22, 24 of the flap assembly 18 are formed in a conventional manner, e.g., by attaching outer skins to one or more lateral structural members. The contours and size of the airfoil elements 22, 24 and the slot 25 will depend on the specific application. At each support location, the elements 22, 24 are mounted atop a pallet 34 having a generally flat upper surface 36. The airfoil elements can be affixed to the pallet upper surface 36 using any one of several known methods, such as welding, riveting, bolting, or the like. For example in FIGURE 1, the pallet includes structural extensions 31, 33 to which the vane and main elements are attached. The chosen mounting method must maintain the airfoil elements in a fixed relation during both stowed and extended positions so as to maintain the slot 25 therebetween. The slot 25 width will depend on the requirements of a specific application as well as the slot's proportions relative to variations in wing chord and wing thickness.

The rear fragment of the wing 30 shown in FIGURES 1-3 includes a spoiler 32 located on the wing upper surface, near the trailing edge. When the slotted cruise trailing edge flap is stowed, the spoiler 32 overlaps a portion of the vane element 22. The spoiler exterior surface is flush with the wing upper skin exterior surface, and the spoiler interior surface is adjacent to the vane element upper surface when the flap assembly 18 is stowed. Therefore, the vane element 22 is 'hidden' beneath the spoiler 32. By positioning the vane element aft edge behind or directly adjacent the spoiler aft edge, the slot 25 will remain open and available to pass air at all times between regions below and above the wing. This arrangement has the additional advantage of allowing the spoiler 32 to be preloaded onto the vane element 22 if need be when the flap assembly is stowed.

At each support location, the extension assembly 20 includes the support structure 23 and the actuation mechanism 28 that moves the flap assembly 18 relative to the support structure 23. Preferably, the flap assembly pallet 34 is rotatably and translatably connected to the support structure 23, while the actuation mechanism 28 is connected directly to the flap assembly pallet 34 in order to force the flap assembly to translate and rotate relative to the support structure 23.

As best seen in FIGURES 2 and 3, the support structure 23 includes a rigid frame 26 having opposed upright side panels 27 oriented in a substantially vertical-longitudinal plane. The upright side panels 27 are cantilevered from the wing rear spar 31. A substantially horizontal track 44 is attached to and between rearwardly extending arms 70 that form part of the opposed frame side panels. Referring to FIGURE 5A, the track 44 is formed of two sub-tracks 45 that each extend inwardly from the upper edge of one frame side panel arm toward the opposite frame side panel arm. The sub-tracks 45 may be formed by forging, milling, bending, etc. A small lateral gap exists between the two sub-

tracks 45.

The support assembly further includes a carriage 46 that engages the track 44 in a manner that permits carriage translation upon the track in a fore and aft direction. In particular, the carriage 46 includes a vertical center plate 47 having four wheels 48 located on each side, the axes of rotation of the wheels being normal to the faces of the vertical center plate 47. As shown in FIGURES 2, 3, and 5A, the vertical center plate 47 is located between the two sub-tracks 45 so that four of the wheels (i.e., two at each side face) contact the upper surfaces of the sub-tracks 45, and four of the wheels contact the lower surfaces of the sub-tracks. The carriage 46 thereby engages the track 44 in a manner that prohibits vertical or lateral disengagement during translation. The carriage is also connected to the flap assembly pallet, as described in detail below.

As shown in FIGURES 2, 3 and 5A, the support structure 23 further includes a cam slot 50 formed along the outer surface of each frame side panel 27. The cam slot 50 acts as a guide that directs the motion of the flap assembly 18 during use. The illustrated cam slot 50 has two portions. The portions are best seen in FIGURES 2 and 3. A first portion 51 runs horizontally and is located near the lower edge of the frame side panels 27. A second portion 53 stems from the aft end of the first portion 51 and forms an upward arc in the side panels 27. More specifically, the second portion 53 angles upwardly and backwardly, in the direction of the spoiler 32, and then in a forward and upward manner. Therefore, rather than being straight, the second portion is curved. The significance of this cam slot shape will be better understood from the description of the motion of the flap assembly during use that is set forth below.

Referring to FIGURE 5A, the pallet 34 includes opposed pallet sidewalls 35 that extend downward from the pallet upper surface 36. The lateral distance between the pallet sidewalls 35 is slightly greater than the lateral distance between the frame side panels 27. This relative sizing permits the pallet sidewalls to straddle the frame side panels 27 in a sleeving manner, where the pallet forms the outer sleeve, and the frame forms the inner sleeve. The pallet 34 is connected to the support structure 23 at two locations on each pallet sidewall 35. The connections are best seen in FIGURES 2 and 3. The first location is at a pivot pin 40 that extends laterally between the pallet sidewalls 35 and through the carriage 46. The pivot pin 40 preferably forms the axle about which the two aftmost carriage upper wheels rotate. The pivot pin location is illustrated in FIGURES 1-3 at a central pivot axis location labeled 42. The second connection location is at the cam slot 50. Each pallet sidewall 35 includes an inwardly-oriented cam follower 52 attached to a lower region of the pallet sidewall. The cam followers 52 are positioned to engage the cam guides 50.

These two connection locations both support the flap assembly 18 and provide a useful method of trans-

lating and rotating the flap assembly 18. In particular, the cam guide first portion 51 first guides the flap assembly to translate aft. When the cam followers 52 enter the cam guide second portion 53, the flap assembly 18 is then caused to rotate downward about the pivot pin 40. It should be noted that, depending on the type of actuation mechanism 28 used to move the flap assembly 18, the cam slot 50 and track 44 must be positioned and shaped to allow smooth translation and rotation of the pallet 34. It is important that the components do not bind or conflict. In this regard, the cam slots may be slightly over-sized to allow the cam followers to move easily within the slots.

Referring back to FIGURE 1, the support structure 23 further includes an outer fairing 60 at each support location that surrounds the frame 26 and the pallet 34. The fairing 60 provides a smooth exterior surface to enhance airflow along the lower surfaces of the wing and the slotted cruise trailing edge flap. The fairing 60 includes a stationary portion 62 connected to the wing lower surface and a movable portion 64 connected to the pallet 34 at the main element 24. The movable portion 64 includes a smaller portion 65 that extends between the vane element 22 and the main element 24 of the flap assembly. This smaller portion 65 closes out the airflow that is passing between the vane and main elements 22, 24.

The actuation mechanism 28 includes a drive apparatus connected to the frame 26 near the wing rear spar 31 at each support location. A rotary actuator 56 is shown in FIGURES 2 and 3, though, other types drives may be used instead, such as linear actuators, etc. Rotational input to the actuators 56 is provided by an independent power source (not shown). Each rotary actuator 56 is preferably positioned to rotate about a laterally-oriented axis. The rotational output of the actuator 56 is passed through a radially extending arm 54 that connects to one end 57 of an idler link 58 in a pin-and-clevis type arrangement. The opposite end 59 of the idler link 58 is rotatably connected to an extension tab 68 connected to a forward region of the pallet 34. The axes of rotation at the idler link ends 57, 59 are parallel with the rotary actuator axis of rotation. These axes are further parallel with the central pivot axis 42.

In moving from a stowed position to an extended position, rotation of the rotary actuator 56 causes the radial arm 54 and idler link 58 to move aft. This action causes the pallet 34 to translate aftward and, thus, the carriage 46 to translate aftward along the track 44. Specifically, the pallet 34 translates aftward a distance commensurate with the linear distance of the track 44 and the cam slot first portion 51. As shown in FIGURE 2, when the cam followers reach the end of the cam guide first portion 51, they continue up the curved second portion 53. The cam followers of FIGURE 2 have started into the second portion 53 and have therefore caused a slight rotation about the central pivot axis at the pivot pin. The angle of rotation shown is about 15 degrees.

As the cam followers 52 move up the cam slot second portions 53, the actuator 56 rotational energy is used to pivot the flap assembly 18 about the central axis. As may be observed in FIGURE 3, very little energy is used past the initial rotation to translate the flap assembly 18 further aft. The flap assembly in FIGURE 3 has been rotated downward about 50 degrees. Other motions may be defined for the flap assembly as required in a particular application by simply manipulating the location and shape of the track 44 and cam slot 50.

As will be appreciated from viewing FIGURES 2 and 3, the extended positions of the above-described flap assembly 18 cause a second slot 66 to form between the spoiler 32 and the vane element 22. This is desirable for those applications requiring a double-slotted flap arrangement. In some applications, though, it may be desirable to limit the amount of aft translation of the flap assembly 18. In such instances, the second slot 66 may be omitted by eliminating the cam slot first portion. The vane flap thereby remains in contact with the spoiler.

While the preferred embodiment of the invention has been illustrated and described, it will be appreciated that various changes can be made therein without departing from the spirit and scope of the invention as defined by the appended claims.

Claims

1. A slotted cruise trailing edge flap apparatus for use on an aircraft wing, the flap apparatus comprising:

- (a) a flap assembly comprising: a main element; a vane element; a pallet, the main and vane elements being mounted on the pallet such that a slot is formed between the main and vane elements; and a cam follower mounted on the pallet; and
- (b) an extension assembly comprising:

- (i) a support structure including a frame connected to the wing, the frame including a cam slot; the support structure further including a carriage translatable engaged with a track, the carriage being rotatably connected to the pallet at a central pivot axis, the cam follower being movably engaged with the cam slot; and
- (ii) an actuation mechanism in pivotable communication with the pallet to cause the pallet to move between stowed and extended positions relative to the support structure, wherein the slot is exposed to airflow during both stowed and extended positions.

2. The slotted cruise trailing edge flap apparatus according to claim 1, wherein the vane and main

elements are mounted to an upper surface of the pallet and the pivotable connection of the pallet to the actuation mechanism is located at a forward region of the pallet near the upper surface; the cam slot is located near a lower region of the pallet; and rotation of the pallet relative to the wing is accomplished about the central pivot axis.

3. The slotted cruise trailing edge flap apparatus according to claim 1 or 2, wherein the track is longitudinally oriented and the carriage includes at least one wheel located below the track and one wheel located above the track.

4. The slotted cruise trailing edge flap apparatus according to claim 3, wherein the at least one wheel located above the track is rotatable about the central pivot axis.

5. The slotted cruise trailing edge flap apparatus according to any of the claims 1-4, further comprising an outer fairing attached to the wing to surround portions of the support structure.

6. The slotted cruise trailing edge flap apparatus according to any of the claims 1-5, wherein the slot between the vane and main elements is a fixed distance.

7. A slotted cruise trailing edge flap apparatus for use on an aircraft wing, the flap apparatus comprising:

- (a) a flap assembly including main and vane elements supported by a pallet; a slot being defined between the main and vane elements; the slot being exposed to airflow during both stowed and extended flap apparatus positions;
- (b) an extension assembly including:

- (i) a support structure attached to the wing at the pallet, the support structure being movably engaged with the pallet, the movable engagement including rotation about a central pivot axis oriented in a generally lateral direction;
- (ii) at least one actuation mechanism connected to the flap assembly to cause the flap assembly to move between stowed and extended positions relative to the support structures, wherein extension of the flap assembly includes rotation about the central pivot axis.

8. The slotted cruise trailing edge flap apparatus according to claim 7, wherein each support structure includes a carriage translatable engaged with a track, the carriage being rotatably connected to the pallet at the central pivot axis; and the movable con-

nection of the flap assembly to the support structure further includes translation of the pallet along the tracks.

9. The slotted cruise trailing edge flap apparatus 5 according to any of the claims 1-4 or 7 or 8, wherein the actuation mechanism includes a rotary actuator pivotably or rotatably linked to the pallet.
10. The slotted cruise trailing edge flap apparatus 10 according to any of the claims 1-4 or 7, 8 or 9, wherein actuation of the rotary actuator further causes the pallet to translate relative to the support structure.
11. The slotted cruise trailing edge flap apparatus 15 according to claim 10, wherein the support structure is shaped to extend the flap assembly aftward prior to rotating the pallet downward.
12. The slotted cruise trailing edge flap apparatus 20 according to any of the claims 1-4 or 7-11, wherein a second slot is formed between the wing and the vane element when the flap apparatus is in an extended position. 25

30

35

40

45

50

55

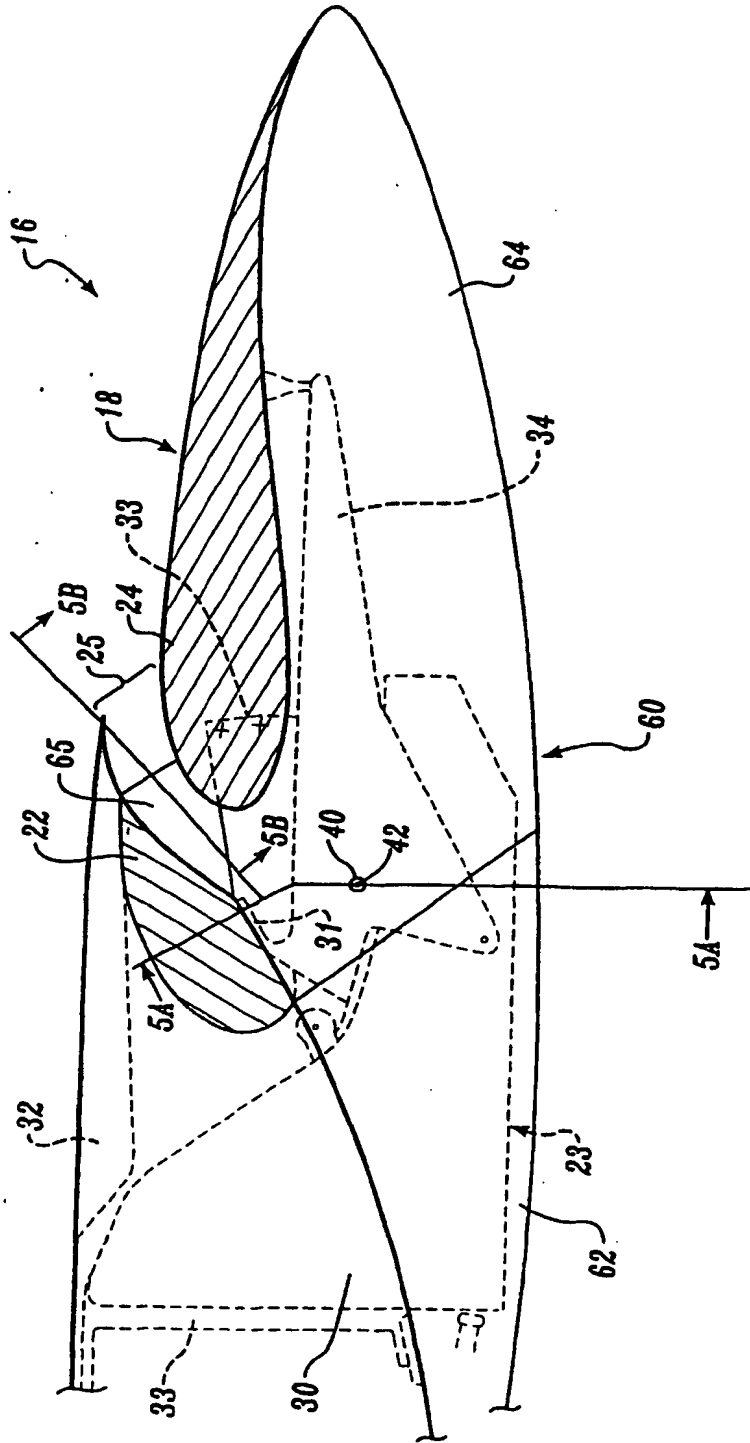


Fig. 1

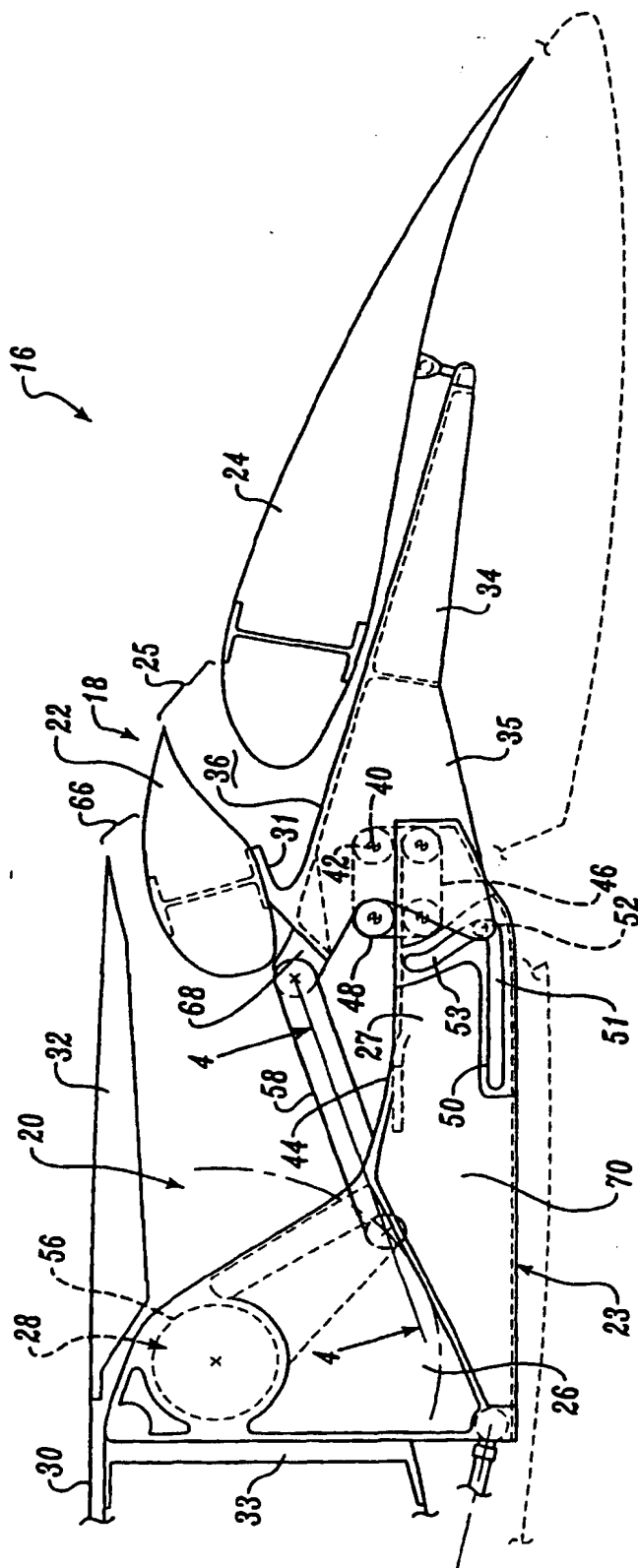


Fig. 2

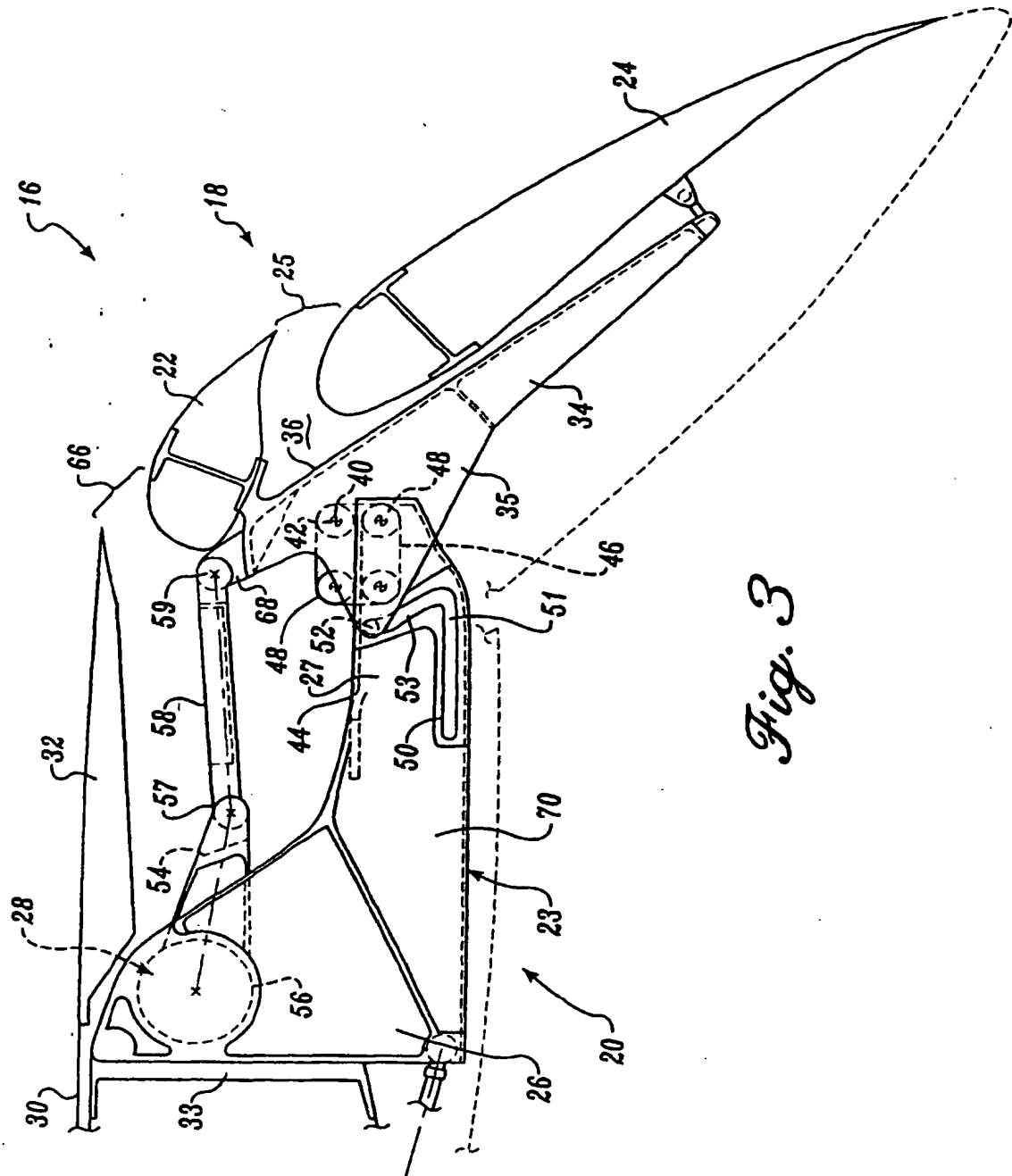


Fig. 3

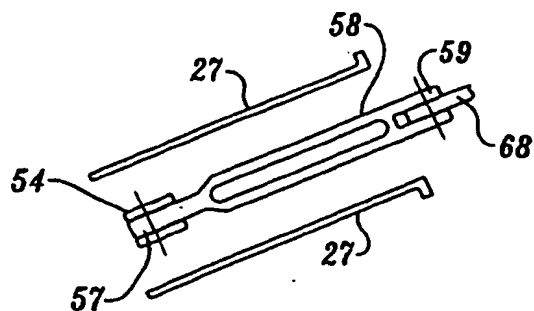


Fig. 4

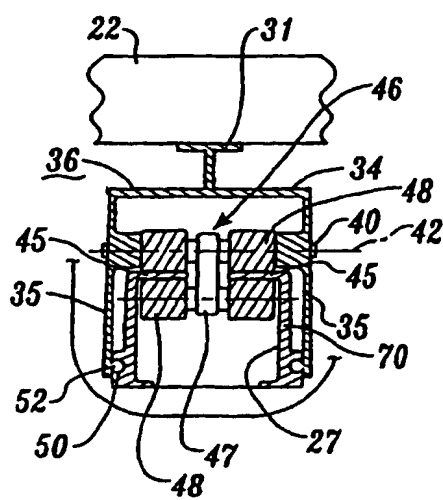


Fig. 5A

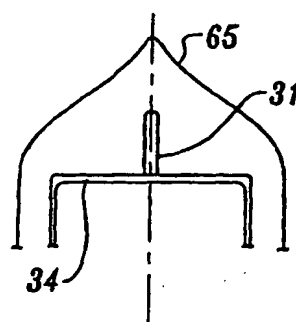
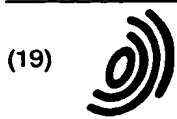


Fig. 5B



(19)

Europäisches Patentamt

European Patent Office

Office européen des brevets



(11)

EP 0 838 394 A3

(12)

EUROPEAN PATENT APPLICATION

(88) Date of publication A3:
21.04.1999 Bulletin 1999/16

(51) Int. Cl.⁶: **B64C 9/02**, B64C 9/16

(43) Date of publication A2:
29.04.1998 Bulletin 1998/18

(21) Application number: 97203214.8

(22) Date of filing: 15.10.1997

(84) Designated Contracting States:
**AT BE CH DE DK ES FI FR GB GR IE IT LI LU MC
NL PT SE**
Designated Extension States:
AL LT LV RO SI

(72) Inventor: **Siers, Geoffrey L.**
Kent, Washinton 98032 (US)

(74) Representative:
**Land, Addick Adrianus Gosling
Arnold & Siedsma,
Advocaten en Octrooigemachtigden,
Sweelinckplein 1
2517 GK Den Haag (NL)**

(30) Priority: 22.10.1996 US 735233

(71) Applicant: **THE BOEING COMPANY**
Seattle, Washington 98124-2207 (US)

(54) Slotted cruise trailing edge flap

(57) A slotted cruise trailing edge flap (16) having a single-slotted configuration during cruise, and a double-slotted configuration during takeoff and landing is disclosed. The slotted cruise trailing edge flap (16) includes a flap assembly (18) positioned along the trailing edge of an aircraft wing (30). The flap assembly (18) includes a vane element (22) and a main element (24) mounted in a fixed relation to one another that defines a slot (25) therebetween. The slotted cruise trailing edge flap also includes an extension assembly (20) that includes a support structure (23) from which the flap assembly (18) is translatably connected via a track (44) and carriage (46) arrangement. The flap assembly (18)

is rotatably connected to the support structure (23) using a cam follower (52) and cam slot (50). An actuation mechanism (28) including a rotary actuator (56) moves the flap assembly (18) relative to the support structure (23). In a stowed position, the vane element (22) preferably nests into the wing (30) such that the slot (25) remains available to direct airflow from the wing lower surface to the wing upper surface. In an extended position, the vane and main elements (22), (24) form a double-slotted arrangement created by the flap assembly translating rearward and rotating downward relative to the wing (30).

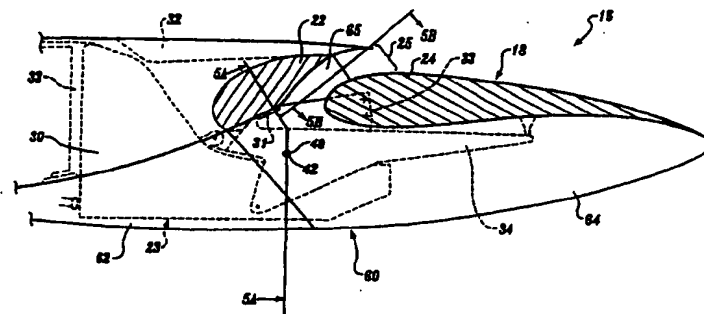


Fig. 1

EP 0 838 394 A3



European Patent
Office

EUROPEAN SEARCH REPORT

Application Number
EP 97 20 3214

DOCUMENTS CONSIDERED TO BE RELEVANT			
Category	Citation of document with indication, where appropriate, of relevant passages	Relevant to claim	CLASSIFICATION OF THE APPLICATION (Int.Cl.6)
Y	US 4 856 735 A (LOTZ DECEASED MICHAEL ET AL) 15 August 1989	1,12	B64C9/02 B64C9/16
A	* column 2. line 61 - column 3, line 31 * * column 3. line 46 - line 56 * * column 4. line 4 - line 6 * * column 5. line 25 - line 35; figure 1B *	6	
Y	US 2 908 454 A (L.J. DE WOLF) 13 October 1959	1,12	
A	* the whole document *	2-4,8, 10,11	
A	EP 0 273 410 A (BOEING CO) 6 July 1988 * figures *	5,9	B64C
A	US 4 784 355 A (BRINE GERALD T) 15 November 1988 * the whole document *	1,5,7,12	
The present search report has been drawn up for all claims			
Place of search THE HAGUE		Date of completion of the search 23 February 1999	Examiner Estrela y Calpe, J
<p>CATEGORY OF CITED DOCUMENTS</p> <p>X : particularly relevant if taken alone Y : particularly relevant if combined with another document of the same category A : technological background O : non-written disclosure P : intermediate document</p> <p>T : theory or principle underlying the invention E : earlier patent document, but published on, or after the filing date D : document cited in the application L : document cited for other reasons & : member of the same patent family, corresponding document</p>			

EPO FORM 1503 03 82 (P/4C01)

EP 0 838 394 A3

ANNEX TO THE EUROPEAN SEARCH REPORT
ON EUROPEAN PATENT APPLICATION NO.

EP 97 20 3214

This annex lists the patent family members relating to the patent documents cited in the above-mentioned European search report.
The members are as contained in the European Patent Office EDP file on
The European Patent Office is in no way liable for these particulars which are merely given for the purpose of information.

23-02-1999

Patent document cited in search report	Publication date	Patent family member(s)	Publication date
US 4856735 A	15-08-1989	DE 3140350 A EP 0076907 A JP 1723139 C JP 3080680 B JP 58076400 A	28-04-1983 20-04-1983 24-12-1992 25-12-1991 09-05-1983
US 2908454 A	13-10-1959	NONE	
EP 0273410 A	06-07-1988	US 4702442 A DE 3584895 A EP 0184230 A	27-10-1987 23-01-1992 11-06-1986
US 4784355 A	15-11-1988	NONE	